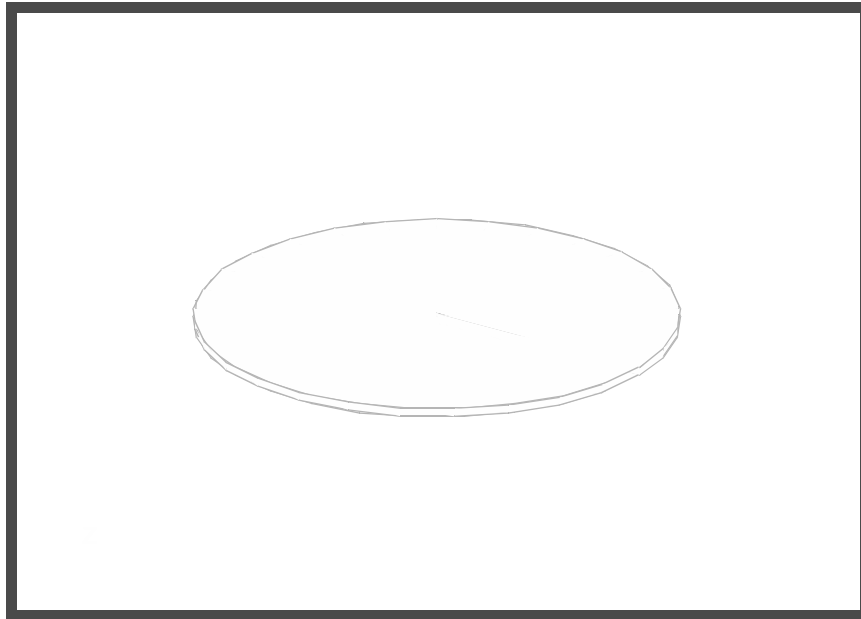

WORKSHOP 6

Modal Analysis of a Circular Plate



Objectives:

- Reduce the model to a 30 degree section and use symmetric boundary conditions.
- Produce a Nastran input file.
- Submit the file for modal analysis in MSC.Nastran.
- Find the first three natural frequencies and mode shapes of the circular plate.



Model Description:

For this example, use Lanczos method to find the first three natural frequencies and mode shapes of a circular plate that is fully clamped around the edge. In addition, model the circular plate by using a 30 degree section to reduce the size of the model. Be certain to incorporate all the necessary symmetric boundary conditions to ensure the accuracy of the analysis.

Below is a finite element representation of a 30° section of the circular plate. It also contains the geometric dimensions and boundary constraints. Table 6.1 contains the necessary parameters to construct the input file (or the model if you are a MSC.Patran user).

Figure 6.1 - Grid Coordinates and Element Connectivities

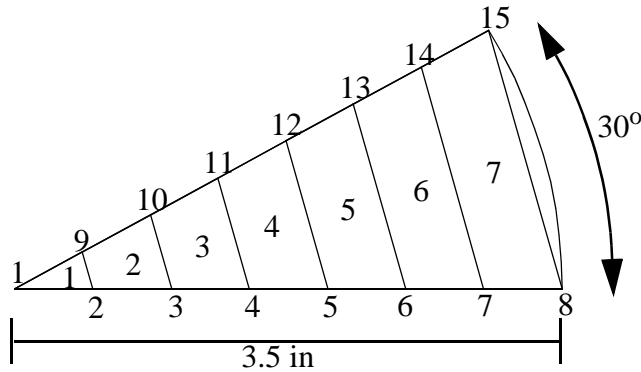


Figure 6.2 - Boundary Constraints

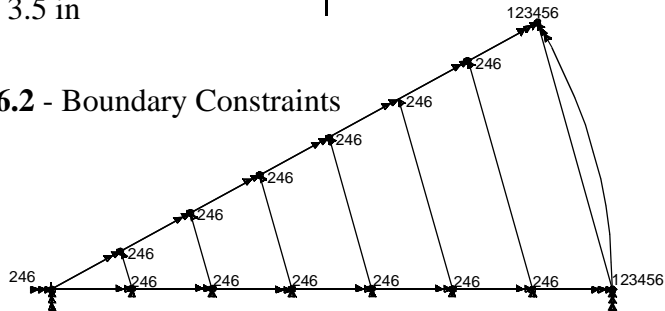


Table 6.1 - Model Properties

Radius:	3.5 in
Thickness:	0.125 in
Weight Density:	0.3 lbs/in³
Mass/Weight Factor:	2.59E-03 sec²/in
Elastic Modulus:	30.0E+06 lbs/in²
Poisson Ratio:	0.3

Theoretical Solution:

Circular flat plate of uniform thickness t and radius r , edge fixed. Uniform load w per unit area including own weight

$$f_n = \frac{K_n}{2\pi} \sqrt{\frac{Dg}{wr^4}}$$

Where	$K_1 = 10.2$	fundamental
	$K_2 = 21.3$	one nodal diameter
	$K_3 = 34.9$	two nodal diameters
	$K_4 = 39.8$	one nodal circle

Notation: f = natural frequency

K_n = constant

g = gravity

E = elastic modulus

$$D = \frac{Et^3}{12(1 - \nu^2)}$$

Reference: Roark and Young. Formula for Stress and Strain, Fifth Edition, McGraw-Hill Book Company

Sample Calculation:

$$D = \frac{30e6(0.125)^3}{12(1 - (0.3)^2)} = 5365.7$$

$$f_n = \frac{10.2}{2\pi} \sqrt{\frac{(5365.7)(386.4)}{(0.3)(0.125)(3.5)^4}} = 985.37Hz$$

Suggested Exercise Steps:

- Explicitly generate a finite element representation of the plate structure i.e., the nodes (GRID) and element connectivity (CQUAD4) should be defined manually.
- Define material (MAT1) and element (PSHELL) properties.
- Apply a clamped boundary constraint (SPC1) to the outer curved edge, and symmetric boundary constraints (SPC1) to the two straight inner edges.
- Prepare the model for a normal modal analysis using Lanczos Method (SOL 103 and PARAMS).
 - PARAM, WTMASS, 0.00259
 - PARAM, COUPMASS, 1
- Generate an input file and submit it to the MSC.Nastran solver for normal modal analysis.
- Review the results, specifically the eigenvalues.

Exercise Procedure:

1. Users who are not utilizing MSC.Patran for generating an input file should go to Step 11. Otherwise, proceed to Step 2.

2. Create a new database called **prob6.db**.

File/New...

New Database Name:

prob6

OK

In the **New Model Preference** form set the following:

Tolerance:

◆ **Default**

Analysis code:

MSC/NASTRAN

OK

In the next few steps, you will create the necessary geometry for the plate.

Whenever possible click **Auto Execute** (turn off).

3. Create a cylindrical coordinate system.

◆ **Geometry**

Action:

Create

Object:

Coord

Method:

3Point

Coord ID List:

1

Type:

Cylindrical

Origin:

[0,0,0]

Point on Axis 3

[0,0,1]

Point on Plane 1-3:

[1,0,0]

Apply

4. Now create a curve.

◆ **Geometry**

Action:

Create

Object:

Curve

Method:

XYZ

Vector Coordinate List:

<3.5, 0, 0>

Origin Coordinate List:

[0, 0, 0]

Apply

5. Create a surface out of the curve you just made.

◆ **Geometry**

Action:

Create

Object:

Surface

Method:

Revolve

Total Angle:

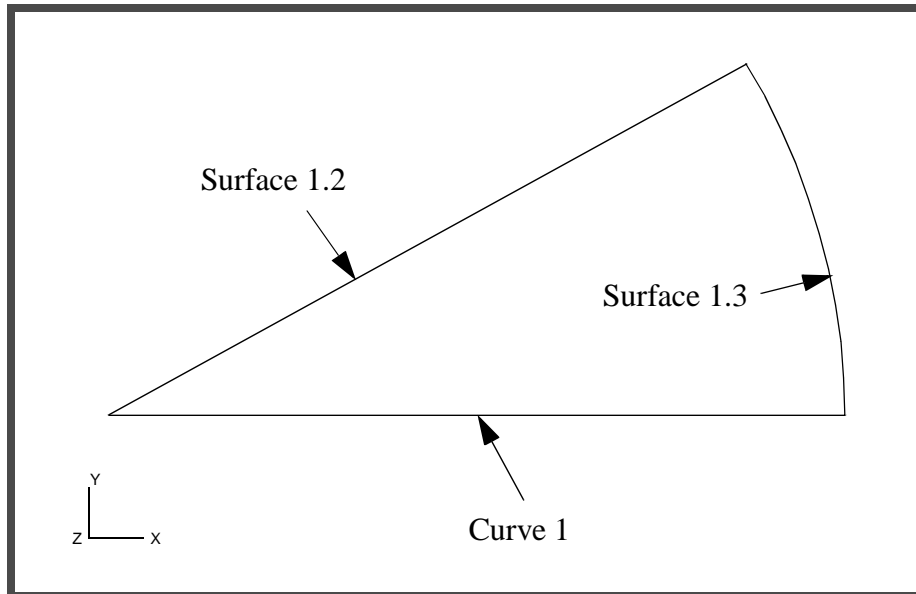
30

Curve List:

Curve 1

Apply

Figure 6.3 - Geometry of Circular Plate



6. Create the Finite Element Model and mesh the surface.

◆ **Finite Elements**

Action:

Create

Object:

Mesh Seed

Type:

Uniform

◆ **Number of Elements**

Number =

7

Curve List:

Curve 1, Surface 1.2

Apply

Now, change the number to 1 and select the right edge.

◆ **Finite Elements**

Action:

Create

Object:

Mesh Seed

Type:

Uniform

◆ **Number of Elements**

Number =
Curve List:

Mesh the surface.

◆ **Finite Elements**

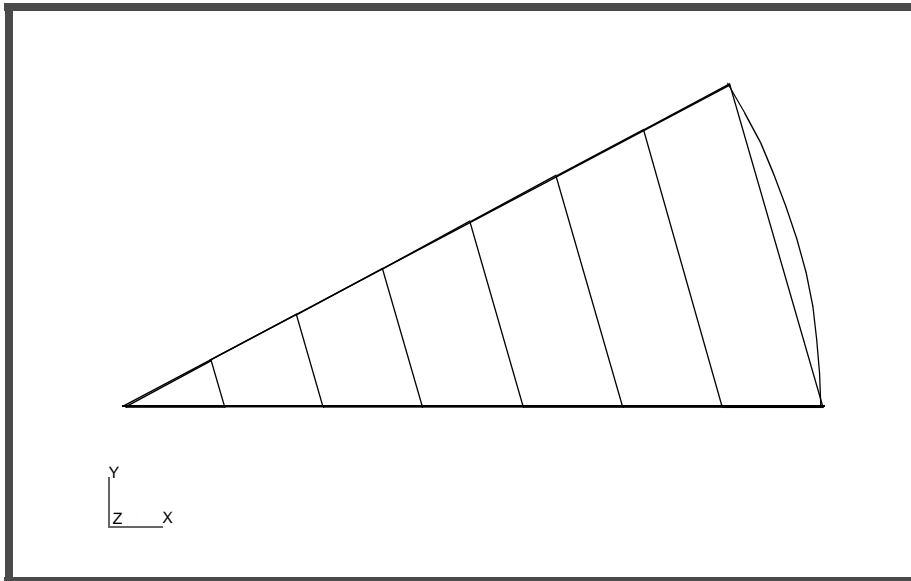
Action:
Object:
Type:

Analysis Coordinate Frame:
Refer. Coordinate Frame:

Surface List:

Click on **OK** if a warning message appears about triangle elements.

Figure 6.4 - Circular Plate Mesh



7. Now you will create the Material Properties for the plate.

◆ **Materials**

Action:

Create

Object:

Isotropic

Method:

Manual Input

Material Name:

mat_1

Input Properties...

Elastic Modulus =

30.0E6

Poisson Ratio =

0.3

Density =

0.3

OK

Apply

8. Give the plate a thickness using Properties.

◆ **Properties**

Action:

Create

<i>Dimension:</i>	<input type="text" value="2D"/>
<i>Type:</i>	<input type="text" value="Shell"/>
<i>Property Set Name:</i>	<input type="text" value="plate"/>
<input type="button" value="Input Properties..."/>	
<i>Material Name:</i>	<input type="text" value="m:mat_1"/>
<i>Thickness:</i>	<input type="text" value="0.125"/>
<input type="button" value="OK"/>	
<i>Select Members:</i>	<input type="text" value="Surface 1"/>
<input type="button" value="Add"/>	
<input type="button" value="Apply"/>	

9. Next you will apply the constraints to the model.

First, constrain the outer edge from moving through all Degrees of Freedom.

◆ **Loads/BCs**

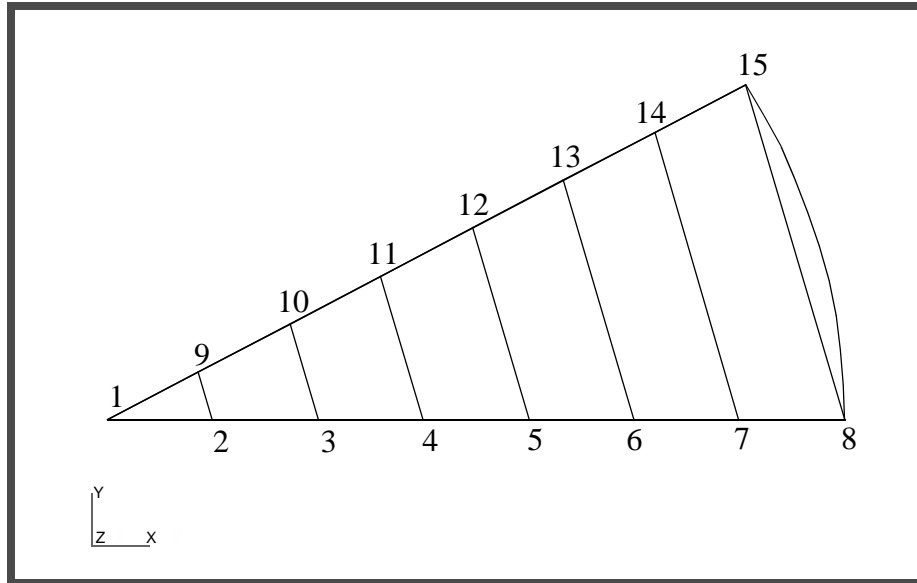
<i>Action:</i>	<input type="text" value="Create"/>
<i>Object:</i>	<input type="text" value="Displacement"/>
<i>Type:</i>	<input type="text" value="Nodal"/>
<i>New Set Name:</i>	<input type="text" value="fixed"/>
<input type="button" value="Input Data..."/>	
<i>Translations <T1 T2 T3></i>	<input type="text" value="<0, 0, 0>"/>
<i>Rotations <R1 R2 R3></i>	<input type="text" value="<0, 0, 0>"/>
<i>Analysis Coordinate Frame:</i>	<input type="text" value="Coord 1"/>
<input type="button" value="OK"/>	

<i>Geometry Filter:</i>	◆ FEM
<i>Select Nodes:</i>	<input type="text" value="Node 8, 15"/>
<i>(see Fig. 6.5)</i>	
<input type="button" value="Add"/>	

OK

Apply

Figure 6.5 - Node IDs



Finally, constrain the upper and lower edge from moving in the Y or θ direction, and from moving in the R1 and R3 rotations.

◆ Loads/BCs

Action:

Create

Object:

Displacement

Type:

Nodal

New Set Name:

symmetry_edge

Input Data...

Translations <T1 T2 T3>

< , 0, >

Rotations <R1 R2 R3>

< 0, , 0>

Analysis Coordinate Frame:

Coord 1

OK

Select Application Region...

Geometry Filter

◆ FEM

Select Nodes

Node 1:7, 9:14

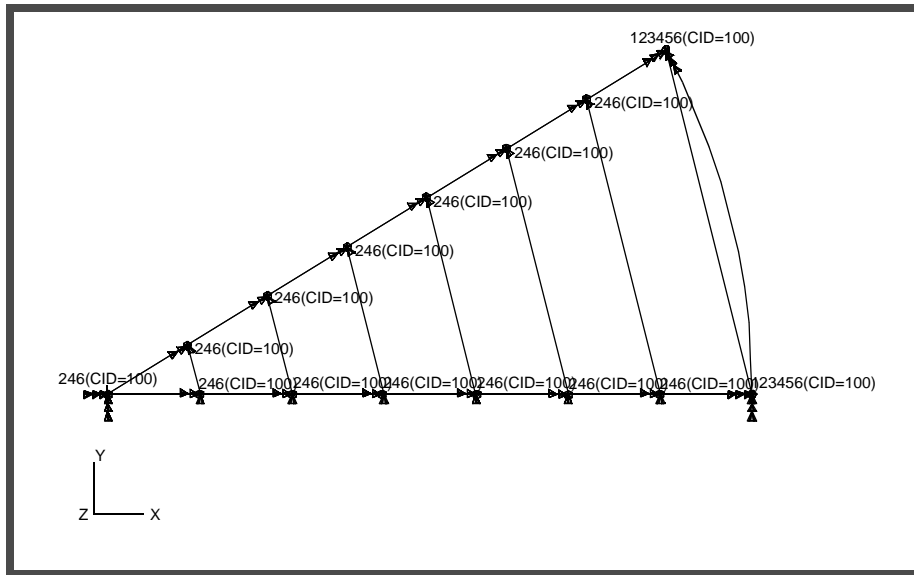
(see Fig. 6.5)

Add

OK

Apply

Figure 6.6 - Nodal Constraints



10. Next, you will run the analysis.

◆ **Analysis**

Action:

Analyze

Object:

Entire Model

Method:

Analysis Deck

Job Name:

prob6

Translation Parameters...

Data Output:

XDB and Print

OK

Solution Type...

Solution Type:

◆ **NORMAL MODES**

Solution Parameters...*Mass Calculation:***Coupled***Wt.-Mass Conversion =***0.00259****OK****OK****Subcase Create...***Available Load Cases:***Default***(Highlight to Select)***Subcase Parameter...***Number of Desired Roots:***3****OK****Apply**

When the message “Do you wish to delete the existing subcase and create a new one” appears, click on *Yes* to create a new subcase.

Cancel**Apply**

An MSC.Nastran input file called **prob6.bdf** will be generated. This process of translating your model into an input file is called the Forward Translation. The Forward Translation is complete when the Heartbeat turns green.

Generating an Input File for MSC.Nastran Users:

MSC.Nastran users can generate an input file using the data from Table 6.1. The result should be similar to the output below.

11. MSC.Nastran Input File: **prob6.dat**.

```
ID SEMINAR,PROB6
TIME 5
SOL 103
CEND
TITLE = FIXED CIRCULAR PLATE
SUBTITLE = NORMAL MODES
    SPC = 20
    METHOD = 1
    DISP = ALL
BEGIN BULK
PARAM, POST, 0
CORD2C,100,,0.0,0.0,0.0,0.0,0.0,1.0
,1.0,0.0,0.0
GRID,1,100,0.0,0.0,0.0,100
=,*1,=,*0.5,==
=6
GRID,9,100,0.5,30.0,0.0,100
=,*1,=,*0.5,==
=5
CTRIA3,1,20,1,2,9
CQUAD4,2,20,2,3,10,9
=,*1,=,*1,*1,*1,*1
=4
PSHELL,20,30,0.125,30,,30
MAT1,30,30.+6,,.3,.3
PARAM,WTMASS,2.59-3
PARAM,COUPMASS,1
SPC,20,8,123456,0.0
SPC,20,15,123456,0.0
SPC1,20,246,1,THRU,8
SPC1,20,246,9,THRU,15
EIGRL,1,,3
ENDDATA
```

Submitting the Input File for Analysis:

12. Submit the input file to MSC.Nastran for analysis.
 - 12a. To submit the MSC.Patran **.bdf** file for analysis, find an available UNIX shell window. At the command prompt enter: **nastran prob6.bdf scr=yes**. Monitor the run using the UNIX **ps** command.
 - 12b. To submit the MSC.Nastran **.dat** file for analysis, find an available UNIX shell window. At the command prompt enter: **nastran prob6.dat scr=yes**. Monitor the run using the UNIX **ps** command.
13. When the run is complete, edit the **prob6.f06** file and search for the word **FATAL**. If no matches exist, search for the word **WARNING**. Determine whether existing **WARNING** messages indicate modeling errors.
14. While still editing truss.f06, search for the word:

E I G E N (spaces are necessary)

1st = _____ Hz

2nd = _____ Hz

3rd = _____ Hz

Comparison of Results:

Compare the results obtained in the **.f06** file with the results on the following page:

REAL EIGENVALUES						
MODE NO.	EXTRACTION ORDER	EIGENVALUE	RADIANS	CYCLES	GENERALIZED MASS	GENERALIZED STIFFNESS
1	1	3.792285E+07	6.158153E+03	9.801005E+02	1.000000E+00	3.792285E+07
2	2	6.074206E+08	2.464590E+04	3.922517E+03	1.000000E+00	6.074206E+08
3	3	3.389181E+09	5.821668E+04	9.265472E+03	1.000000E+00	3.389181E+09

15. **MSC.Nastran Users have finished this exercise. MSC.Patran Users should proceed to the next step.**

16. Proceed with the Reverse Translation process; that is, attaching the **prob6.xdb** results file into MSC.Patran. To do this, return to the **Analysis** form and proceed as follows:

◆ **Analysis**

Action:

Attach XDB

Object:

Result Entities

Method:

Local

Select Results File...

Selected Results File:

prob6.xdb

OK

Apply

17. When the translation is complete bring up the **Results** form.

◆ **Results**

Action:

Create

Object:

Deformation

Select Result Case(s):

Default, A1:Mode 1:Freq. = 980.1

Select Deformation Result:

Eigenvectors, Translational

Show As:

Resultant

Apply

Repeat for mode 2 and 3

You may reset the graphics if you click on this icon:



Reset Graphics

You can go back and select any *Results Case, Fringe Results or Deformation Results* you are interested in.

Quit MSC.Patran when you are finished with this exercise.