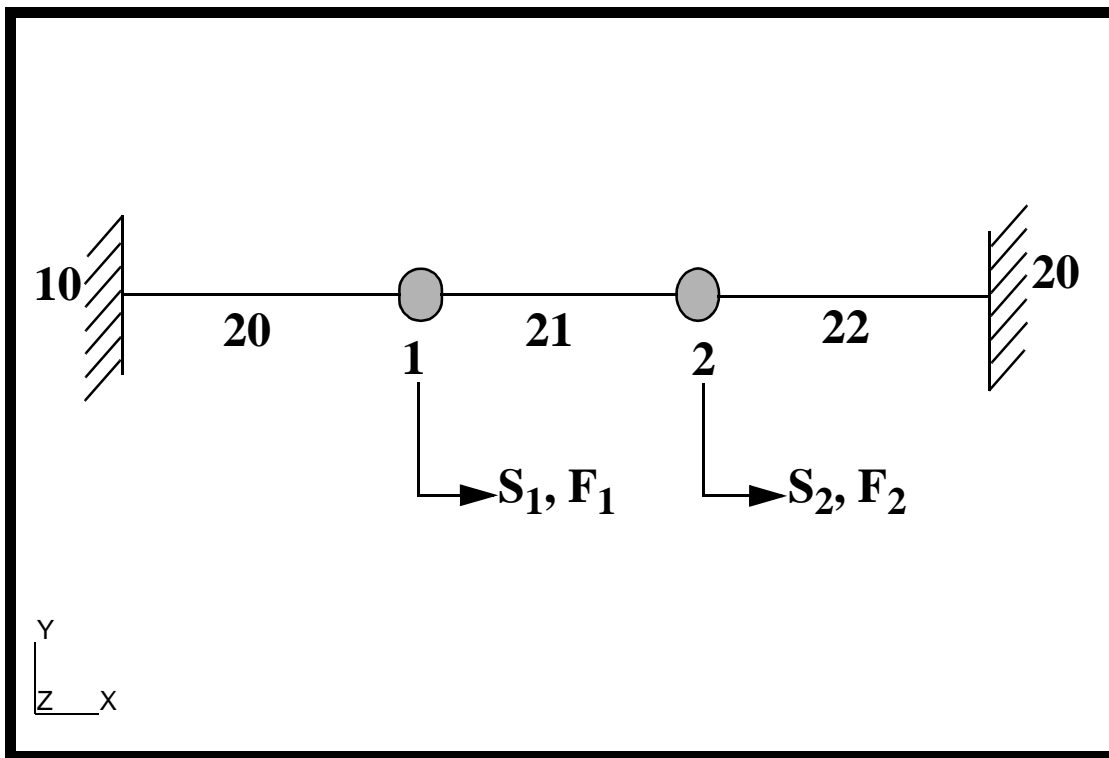


## WORKSHOP 7

### Objectives:

- Multi-disciplinary optimization problem subject to static, normal modes, and frequency response.





## Model Description:

- Subject to:
  - Static loads of  $S_1 = 30$  and  $S_2 = 40$  pounds at Grid Points 1 and 2, respectively.
  - Normal modes analysis.
  - $F_1 = F_2 = 10$  pounds harmonic load at Grid Points 1 and 2.

- Objective Function - Minimize the SRSS response

$$\sqrt{(y_{f1}^2 + y_{f2}^2 + y_{f2}^2 + \dots)}$$

of Grid Point 2 from 4 hz to 30 hz.

- Design Variables: cross-sectional area A1, A2, and A3

- Constraints:

Static subcase:  $-20,000 < \text{Elem stresses} < +20,000$

Modal subcase:  $17 \text{ hz} < \text{Freq1} < 20 \text{ hz}$

---

# Generating an input file for MSC.Nastran Users:

1. Generate an input file using the data from pages 7-1 through 7-3. Use the following input file as a starting point.

```
$
$   wkshp7.dat
$
TIME 5
SOL 200 $
CEND
TITLE=TWO MASS OPTIMIZATION PROBLEM
DISPL = ALL
$
SUBCASE 1
  LABEL = STATIC
  ANALYSIS = ....
  LOAD = 101
  FORCE = ALL
  STRESS = ALL
  DESSUB = ....
$
SUBCASE 2
  LABEL = MODES
  ANALYSIS = ....
  METHOD = 201
  DESSUB = ....
$
SUBCASE 3
  LABEL = DIRECT FREQUENCY RESPONSE
  ANALYSIS = ....
  DISP(PHASE)=ALL
  FREQ = 100
  DLOAD = 200
  DESOBJ(MIN) = ....
$
BEGIN BULK
$
param,post,-1
$
FORCE 101 1 30. 1.
FORCE 101 2 40. 1.
EIGRL 201 2
$
CROD 20 20 10 1
CROD 21 21 1 2
CROD 22 22 2 20
CONM2 10 1 1.
CONM2 11 2 2.
DAREA 201 1 1 10. 2 1 10.
FREQ1 100 4. .25 30
GRID 10 0. 0. 0. 123456
GRID 1 1. 0. 0. 23456
GRID 2 2. 0. 0. 23456
GRID 20 3. 0. 0. 123456
PROD 20 200 1.E-3
PROD 21 200 1.5E-3
PROD 22 200 2.E-3
MAT1,200,1.0E7,,0.3,1.0 $ DUMMY DENSITY
RLOAD1 200 201 210
TABLED1 210
  1. 1. 10. 1. ENDT
param,g,.10
$
$   design model
$
DOPTPRM P1 1 P2 15 DESMAX 10
.
.
ENDDATA
```

## WORKSHOP 7

2. The completed MSC.Nastran input file is shown below:

```

$
$   soln7.dat
$
TIME 5
SOL 200 $
CEND
TITLE=TWO MASS OPTIMIZATION PROBLEM
DISPL = ALL
$
SUBCASE 1
  LABEL = STATIC
  ANALYSIS = STATICS
  LOAD = 101
  FORCE = ALL
  STRESS = ALL
  DESSUB = 100
$
SUBCASE 2
  LABEL = MODES
  ANALYSIS = MODES
  METHOD = 201
  DESSUB = 200
$
SUBCASE 3
  LABEL = DIRECT FREQUENCY RESPONSE
  ANALYSIS = DFREQ
  DISP(PHASE)=ALL
  FREQ = 100
  DLOAD = 200
DESOBJ(MIN) = 1000
$
BEGIN BULK
$
param,post,-1
$
FORCE  101    1           30.    1.
FORCE  101    2           40.    1.
EIGRL  201           2
$
CROD   20     20     10     1
CROD   21     21     1     2
CROD   22     22     2     20
CONM2  10     1           1.
CONM2  11     2           2.
DAREA  201    1           1     10.    2     1     10.
FREQ1  100    4.     .25    104
GRID   10           0.     0.     0.     123456
GRID   1           1.     0.     0.     23456
GRID   2           2.     0.     0.     23456
GRID   20          3.     0.     0.     123456
PROD   20     200     1.E-3
PROD   21     200     1.5E-3
PROD   22     200     1.5E-3
MAT1,200,1.0E7,,0.3,1.0 $ DUMMY DENSITY
RLOAD1 200     201
TABLED1 210
  1.     1.     10.    1.     ENDT
param,g,.10
$
$   design model
$
DOPTPRM P1     1     P2     15     DESMAX  10
DESVAR  1     A1     1.-3    1.-4    1.-2
DESVAR  2     A2     1.5-3    1.5-4    1.5-2
DESVAR  3     A3     2.-3    2.0-4    2.0-2
DVPREL1 1     PROD    20     4     1.0-3
  1     1.0
DVPREL1 2     PROD    21     4     1.5-3
  2     1.0

```

```

DVPREL1 3      PROD    22      4      2.0-3
          3      1.0
DCONSTR 100    101     -20000. 20000.
DCONSTR 100    102     -20000. 20000.
DCONSTR 100    103     -20000. 20000.
DCONSTR 200    200      17.     20.
$
DRESP1  101     STRELM1 STRESS  PROD          2          20
DRESP1  102     STRELM2 STRESS  PROD          2          21
DRESP1  103     STRELM3 STRESS  PROD          2          22
DRESP1  200     MODE1   FREQ     1
$
dresp2,1000,srssg2c1,2000
,dresp1,1001,1002,1003,1004,1005,1006,1007
,,*7,*7,*7,*7,*7,*7,*7,*7
=13
$
DRESP1,1001,GRID2C1,FRDISP,,,1,4.0,2
=,*1,=,=,=,=,=,*(.25),=
=103
$
deqatn  2000    srssg2c1(f1,f2,f3,f4,f5,f6,f7,f8,f9,f10,f11,f12,f13,f14,
                      f15,f16,f17,f18,f19,f20,f21,f22,f23,f24,
                      f25,f26,f27,f28,f29,f30,f31,f32,f33,f34,
                      f35,f36,f37,f38,f39,f40,f41,f42,f43,f44,
                      f45,f46,f47,f48,f49,f50,f51,f52,f53,f54,
                      f55,f56,f57,f58,f59,f60,f61,f62,f63,f64,
                      f65,f66,f67,f68,f69,f70,f71,f72,f73,f74,
                      f75,f76,f77,f78,f79,f80,f81,f82,f83,f84,
                      f85,f86,f87,f88,f89,f90,f91,f92,f93,f94,
                      f95,f96,f97,f98,f99,f100,
                      f101,f102,f103,f104,f105) =
sqrt(f1**2+f2**2+f3**2+f4**2+f5**2+f6**2+f7**2+
f8**2+f9**2+f10**2+f11**2+f12**2+f13**2+f14**2+
f15**2+f16**2+f17**2+f18**2+f19**2+f20**2+
f21**2+f22**2+f23**2+f24**2+f25**2+f26**2+
f27**2+f28**2+f29**2+f30**2+f31**2+f32**2+
f33**2+f34**2+f35**2+f36**2+f37**2+f38**2+
f39**2+f40**2+f41**2+f42**2+f43**2+f44**2+
f45**2+f46**2+f47**2+f48**2+f49**2+f50**2+
f51**2+f52**2+f53**2+f54**2+f55**2+f56**2+
f57**2+f58**2+f59**2+f60**2+f61**2+f62**2+
f63**2+f64**2+f65**2+f66**2+f67**2+f68**2+
f69**2+f70**2+f71**2+f72**2+f73**2+f74**2+
f75**2+f76**2+f77**2+f78**2+f79**2+f80**2+
f81**2+f82**2+f83**2+f84**2+f85**2+f86**2+
f87**2+f88**2+f89**2+f90**2+f91**2+f92**2+
f93**2+f94**2+f95**2+f96**2+f97**2+f98**2+
f99**2+f100**2+
f101**2+f102**2+f103**2+f104**2+f105**2)
ENDDATA

```

3. Submit the input file to MSC.Nastran for analysis.

To submit the MSC.Nastran **.dat** file, find an available UNIX shell window and at the command prompt enter **nastran wkshp7 scr=yes**. Monitor the run using the UNIX **ps** command.

4. When the run is completed, edit the **wkshp7.f06** file and search for the word **FATAL**. If no matches exist, search for the word **WARNING**. Determine whether existing **WARNING** messages indicate modeling errors.

- 4a. While still editing **wkshp7.f06**, search for the word:

H I S T O R Y

# Comparison of Results:

5. Compare the results obtained in the .f06 file with the following:

\*\*\*\*\*  
SUMMARY OF DESIGN CYCLE HISTORY  
\*\*\*\*\*

(HARD CONVERGENCE ACHIEVED)  
(SOFT CONVERGENCE ACHIEVED)

NUMBER OF FINITE ELEMENT ANALYSES COMPLETED 5  
NUMBER OF OPTIMIZATIONS W.R.T. APPROXIMATE MODELS 4

OBJECTIVE AND MAXIMUM CONSTRAINT HISTORY

CYCLE NUMBER	OBJECTIVE FROM APPROXIMATE OPTIMIZATION	OBJECTIVE FROM EXACT ANALYSIS	FRACTIONAL ERROR OF APPROXIMATION	MAXIMUM VALUE OF CONSTRAINT
INITIAL		2.092793E-02		2.692307E-01
1	2.326684E-02	1.880714E-02	2.371285E-01	-1.848973E-02
2	1.865978E-02	1.833768E-02	1.756471E-02	-1.624980E-02
3	1.790566E-02	1.744772E-02	2.624666E-02	-3.792715E-02
4	1.744772E-02	1.744772E-02	0.000000E+00	-3.792715E-02

1 TWO MASS OPTIMIZATION PROBLEM AUGUST 14, 1998 MSC/NASTRAN 4/28/98 PAGE 308

0

DESIGN VARIABLE HISTORY

INTERNAL DV. ID.	EXTERNAL DV. ID.	LABEL	INITIAL	1	2	3	4	5
1	1	A1	1.0000E-03	1.5665E-03	1.4401E-03	1.2765E-03	1.2765E-03	
2	2	A2	1.5000E-03	1.5968E-03	1.7829E-03	4.6500E-03	4.6500E-03	
3	3	A3	2.0000E-03	2.0615E-03	2.2233E-03	2.5143E-03	2.5143E-03	

SUBCASE 2

# X-Y Plots of Design Results:

Figure 7.1 - Objective Function

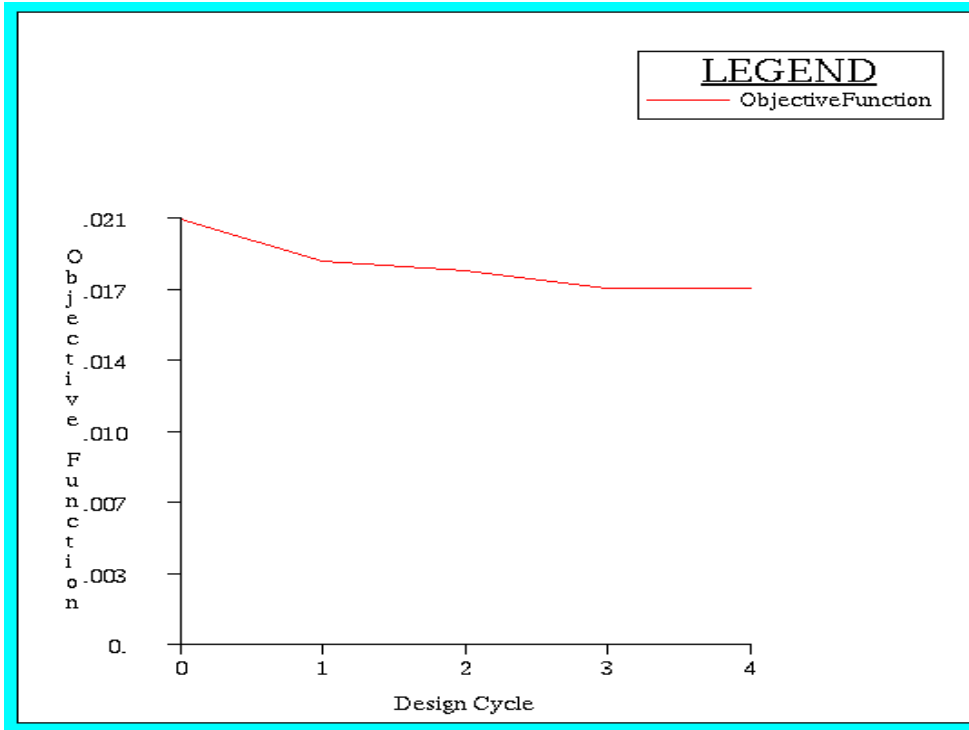
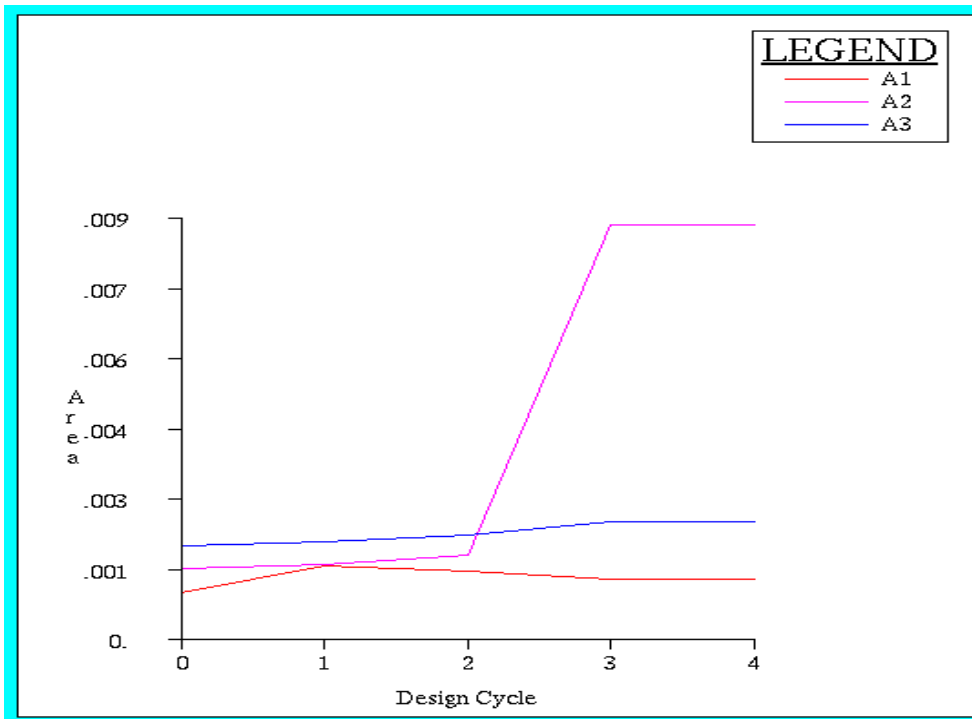


Figure 7.2 - Design Variables



**Figure 7.3 - Maximum Constraint**

